



A Computerized Boundary Element Algorithm for Modeling and Optimization of Complex Magneto-Thermoelastic Problems in MFGA Structures

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Authors' contributions

This work was carried out in collaboration between all authors. All authors read and approved the final manuscript.

Article Information

DOI: 10.9734/JERR/2018/v3i216872

Editor(s):

(1) Dr. Anuj Kumar Goel, Associate Professor, Maharishi Markandeshwar University, (Deemed to be University), Ambala, India.

Reviewers:

(1) Francisco Bulnes, Technological Institute of High Studies of Chalco, Mexico.

(2) Jungki Lee, Hongik University, South Korea.

Complete Peer review History: <http://www.sciencedomain.org/review-history/28098>

Original Research Article

Received 10 October 2018
Accepted 17 December 2018
Published 03 January 2019

ABSTRACT

Aims: The aim of this article is to propose a boundary integral equation algorithm for modeling and optimization of magneto-thermoelastic problems in multilayered functionally graded anisotropic (MFGA) structures.

Study Design: Original research paper.

Place and Duration of Study: Jamoum laboratory, January 2018.

Methodology: a new dual reciprocity boundary element algorithm was implemented for solving the governing equations of magneto-thermoelastic problems in MFGA structures.

Results: A numerical results demonstrate validity, accuracy, and efficiency of the presented technique.

Conclusion: Our results thus confirm the validity, accuracy, and efficiency of the proposed technique. It is noted that the obtained dual reciprocity boundary element method (DRBEM) results are more accurate than the FEM results, the DRBEM is more efficient and easy to use than FEM because it only needs the boundary of the domain needs to be discretized.

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Keywords: Predictor-corrector; shape optimization; magneto-thermoelasticity; functionally graded anisotropic structures; dual reciprocity boundary element method.

2010 mathematics subject classification: 65M38 - 65K05 - 74B05 - 74E05 - 74F05 - 74H05 - 74H15 - 74S20 - 90C31.

1. INTRODUCTION

An understanding of behaviour of functionally graded anisotropic magneto-thermoelastic materials has great practical applications in applied sciences and engineering. In recent years, many researchers discussed the behavior of MFGA structures. With the new advances in computer hardware and software, it is now possible to solve complex magneto-thermoelastic problems by using the DRBEM. which proposed by Nardini and Brebbia [1]. The interested readers can find more details in the following references [2-6].

The aim of this article is to propose a new DRBEM algorithm for solving the governing equations of magneto-thermoelastic problems in MFGA structures. The obtained numerical results demonstrate validity, accuracy, and efficiency of the proposed technique.

2. FORMULATION OF THE PROBLEM

Consider a MFGA structure occupies the region $R = \{(x, y, z): 0 < x < h, 0 < y < b, 0 < z < a\}$. At each and every point on the boundary C , the temperature and displacement are suitably specified.

According to Green and Naghdi theory, the governing equations of MFGA structures for the i th layer, can be expressed as [7-10]:

$$\sigma_{ab,b} + \tau_{ab,b} = \rho^i(x+1)^m \ddot{u}_a^i \quad (1)$$

$$\sigma_{ab} = (x+1)^m [C_{abfg}^i u_{f,g}^i - \beta_{ab}^i (T^i - T_0 + \tau_1 \dot{T}^i)] \quad (2)$$

$$\tau_{ab} = \mu^i(x+1)^m (\tilde{h}_a H_b + \tilde{h}_b H_a - \delta_{ba} (\tilde{h}_f H_f)) \quad (3)$$

$$\left[\left(k_{ab}^{i*} + k_{ab}^i \frac{\partial}{\partial \tau} \right) T_{,ab}^i + \rho^i \dot{\mathfrak{X}} \right] = \beta_{ab}^i T_0 \ddot{u}_{a,b}^i + \rho^i c^i (x+1)^m \dot{T}^i \quad (4)$$

where σ_{ab} , τ_{ab} , u_k^i and T^i are respectively mechanical stress, τ_{ab} Maxwell's stress,

displacement and temperature, T_0 , C_{abfg}^i , β_{ab}^i , μ^i , \tilde{h} , k_{ab}^i , k_{ab}^{i*} , ρ^i and τ are respectively reference temperature, constant elastic moduli, stress-temperature coefficients, magnetic permeability, perturbed magnetic field, thermal conductivity coefficients, new material coefficients associated with the GN theories, density and time, c^i is the specific heat capacity, τ_1 is the relaxation times, \mathfrak{X} is the heat source, ; $i = 1, 2, \dots, n-1$. represents the parameters in multilayered plate, respectively, and $f(x)$ is a given nondimensional function of space variable x . We take $f(x) = (x+1)^m$, where m is a dimensionless constant.

3. DRBEM IMPLEMENTATION

Using the same technique of Fahmy [11-13] for the current problem and implementing the DRBEM, we can write the boundary integral representation formula of coupled thermoelasticity as follows:

$$U_D^i(\xi) = \int_C (U_{DA}^{i*} T_A^i - \check{T}_{DA}^{i*} U_A^i) dC + \sum_{q=1}^E \left(U_{DE}^{iq}(\xi) + \int_C (T_{DA}^{i*} U_{AE}^{iq} - U_{DA}^{i*} T_{AE}^{iq}) dC \right) \alpha_E^q \quad (5)$$

According to Fahmy [14], the DRBEM equation (5) can be written as:

$$\check{\mathfrak{Z}} U - \eta T = (\check{\mathfrak{Z}} \check{U} - \eta \check{\mathfrak{Z}} \check{\mathfrak{D}}) \alpha \quad (6)$$

An implicit-implicit staggered algorithm based on DRBEM was implemented for solving the governing equations which can be written using (6) as follows:

$$\check{M} \check{U}^i + \check{F} \check{U}^i + \check{K} U^i = \check{Q}^i \quad (7)$$

$$\check{\mathfrak{X}} \check{T}^i + \check{A} \check{T}^i + \check{B} T^i = \check{Z} \check{U}^i + \check{R} \quad (8)$$

where the matrices in (7) and (8) are as follows:

$$\begin{aligned} V &= (\eta\tilde{\varphi} - \zeta\tilde{U})J^{-1}, & \tilde{M} &= V(\bar{A} + \bar{\delta}_{AF}), & \tilde{\Gamma} &= V\bar{\Gamma}_{AF}, \\ \tilde{K} &= -\zeta + V(B^T J^{-1} + \bar{\psi}), & \tilde{Q}^i &= -\eta\tilde{T} + VS^0, & \tilde{X} &= -\rho^i c^i (x + 1)^m, \\ \tilde{A} &= k_{ab}^i \frac{\partial}{\partial x_a} \frac{\partial}{\partial x_b}, & \tilde{B} &= k_{ab}^{i*} \frac{\partial}{\partial x_a} \frac{\partial}{\partial x_b}, & \tilde{Z} &= \beta_{ab}^i T_0, \\ \tilde{R} &= -\rho^i \dot{x}. \end{aligned}$$

Equations (7) and (8) yield the following system [15]:

$$\tilde{M} \dot{U}_{n+1}^i + \tilde{\Gamma} \dot{U}_{n+1}^i + \tilde{K} U_{n+1}^i = \tilde{Q}_{n+1}^{ip} \quad (9)$$

$$\tilde{X} \dot{T}_{n+1}^i + \tilde{A} T_{n+1}^i + \tilde{B} T_{n+1}^i = \tilde{Z} \dot{U}_{n+1}^i + \tilde{R} \quad (10)$$

where $\tilde{Q}_{n+1}^{ip} = \eta T_{n+1}^{ip} + VS^0$ and T_{n+1}^{ip} is the predicted temperature.

Integrating Eq. (7) and using Eq. (9), we get

$$\begin{aligned} \dot{U}_{n+1}^i &= \dot{U}_n^i + \frac{\Delta\tau}{2} (\dot{U}_{n+1}^i + \dot{U}_n^i) \\ &= \dot{U}_n^i + \frac{\Delta\tau}{2} \left[\dot{U}_n^i + \tilde{M}^{-1} \left(\tilde{Q}_{n+1}^{ip} - \tilde{\Gamma} \dot{U}_{n+1}^i - \tilde{K} U_{n+1}^i \right) \right] \end{aligned} \quad (11)$$

$$\begin{aligned} U_{n+1}^i &= U_n^i + \frac{\Delta\tau}{2} (\dot{U}_{n+1}^i + \dot{U}_n^i) \\ &= U_n^i + \Delta\tau \dot{U}_n^i + \frac{\Delta\tau^2}{4} \left[\dot{U}_n^i + \tilde{M}^{-1} \left(\tilde{Q}_{n+1}^{ip} - \tilde{\Gamma} \dot{U}_{n+1}^i - \tilde{K} U_{n+1}^i \right) \right] \end{aligned} \quad (12)$$

From Eq. (11) we have

$$\dot{U}_{n+1}^i = \bar{Y}^{-1} \left[\dot{U}_n^i + \frac{\Delta\tau}{2} \left[\dot{U}_n^i + \tilde{M}^{-1} \left(\tilde{Q}_{n+1}^{ip} - \tilde{K} U_{n+1}^i \right) \right] \right] \quad (13)$$

where $\bar{Y} = \left(I + \frac{\Delta\tau}{2} \tilde{M}^{-1} \tilde{\Gamma} \right)$

Substituting from Eq. (13) into Eq. (12), we derive

$$\begin{aligned} U_{n+1}^i &= U_n^i + \Delta\tau \dot{U}_n^i + \frac{\Delta\tau^2}{4} \\ &\left[\dot{U}_n^i + \tilde{M}^{-1} \left(\tilde{Q}_{n+1}^{ip} - \tilde{\Gamma} \bar{Y}^{-1} \left[\dot{U}_n^i + \frac{\Delta\tau}{2} \left[\dot{U}_n^i + \tilde{M}^{-1} \left(\tilde{Q}_{n+1}^{ip} - \tilde{K} U_{n+1}^i \right) \right] \right] - \tilde{K} U_{n+1}^i \right) \right] \end{aligned} \quad (14)$$

Substituting \dot{U}_{n+1}^i from Eq. (13) into Eq. (9) we obtain

$$\dot{U}_{n+1}^i = \tilde{M}^{-1} \left[\tilde{Q}_{n+1}^{ip} - \tilde{\Gamma} \left[\bar{Y}^{-1} \left[\dot{U}_n^i + \frac{\Delta\tau}{2} \left[\dot{U}_n^i + \tilde{M}^{-1} \left(\tilde{Q}_{n+1}^{ip} - \tilde{K} U_{n+1}^i \right) \right] \right] \right] - \tilde{K} U_{n+1}^i \right] \quad (15)$$

Integrating Eq. (8) and using Eq. (10) we have

$$\begin{aligned} \dot{T}_{n+1}^i &= \dot{T}_n^i + \frac{\Delta\tau}{2} (\dot{T}_{n+1}^i + \dot{T}_n^i) \\ &= \dot{T}_n^i + \frac{\Delta\tau}{2} \left(\tilde{X}^{-1} \left[\tilde{Z} \dot{U}_{n+1}^i + \tilde{R} - \tilde{A} T_{n+1}^i - \tilde{B} T_{n+1}^i \right] + \dot{T}_n^i \right) \end{aligned} \quad (16)$$

$$\begin{aligned}
 T_{n+1}^i &= T_n^i + \frac{\Delta\tau}{2}(\dot{T}_{n+1}^i + \dot{T}_n^i) \\
 &= T_n^i + \Delta\tau\dot{T}_n^i + \frac{\Delta\tau^2}{4}\left(\dot{T}_n^i + \tilde{\mathcal{X}}^{-1}\left[\tilde{\mathcal{Z}}\dot{U}_{n+1}^i + \tilde{\mathcal{R}} - \tilde{\mathcal{A}}\dot{T}_{n+1}^i - \tilde{\mathcal{B}}T_{n+1}^i\right]\right)
 \end{aligned}
 \tag{17}$$

From Eq. (16) we get

$$\dot{T}_{n+1}^i = \mathcal{Y}^{-1}\left[\dot{T}_n^i + \frac{\Delta\tau}{2}\left(\tilde{\mathcal{X}}^{-1}\left[\tilde{\mathcal{Z}}\dot{U}_{n+1}^i + \tilde{\mathcal{R}} - \tilde{\mathcal{B}}T_{n+1}^i\right] + \dot{T}_n^i\right)\right]
 \tag{18}$$

where $\mathcal{Y} = \left(I + \frac{1}{2}\tilde{\mathcal{A}}\Delta\tau\tilde{\mathcal{X}}^{-1}\right)$

Substituting from Eq. (18) into Eq. (17), we have

$$\begin{aligned}
 T_{n+1}^i &= T_n^i + \Delta\tau\dot{T}_n^i + \frac{\Delta\tau^2}{4}\left(\dot{T}_n^i + \tilde{\mathcal{X}}^{-1}\left[\tilde{\mathcal{Z}}\dot{U}_{n+1}^i + \tilde{\mathcal{R}}\right.\right. \\
 &\quad \left.\left.- \tilde{\mathcal{A}}\left(\mathcal{Y}^{-1}\left[\dot{T}_n^i + \frac{\Delta\tau}{2}\left(\tilde{\mathcal{X}}^{-1}\left[\tilde{\mathcal{Z}}\dot{U}_{n+1}^i + \tilde{\mathcal{R}} - \tilde{\mathcal{B}}T_{n+1}^i\right] + \dot{T}_n^i\right)\right)\right] - \tilde{\mathcal{B}}T_{n+1}^i\right)\right)
 \end{aligned}
 \tag{19}$$

Substituting \dot{T}_{n+1}^i from Eq. (18) into Eq. (10) we obtain

$$\begin{aligned}
 \ddot{T}_{n+1}^i &= \tilde{\mathcal{X}}^{-1}\left[\tilde{\mathcal{Z}}\dot{U}_{n+1}^i + \tilde{\mathcal{R}}\right. \\
 &\quad \left.- \tilde{\mathcal{A}}\left(\mathcal{Y}^{-1}\left[\dot{T}_n^i + \frac{\Delta\tau}{2}\left(\tilde{\mathcal{X}}^{-1}\left[\tilde{\mathcal{Z}}\dot{U}_{n+1}^i + \tilde{\mathcal{R}} - \tilde{\mathcal{B}}T_{n+1}^i\right] + \dot{T}_n^i\right)\right)\right] - \tilde{\mathcal{B}}T_{n+1}^i\right]
 \end{aligned}
 \tag{20}$$

Using the algorithm of Fahmy [16-22], we have the temperature and the displacements.

4. SHAPE DESIGN SENSITIVITY ANALYSIS AND OPTIMIZATION

Thus, the design sensitivities with respect to the design variables x_h for the displacement and temperature which describe the structural response are performed by implicit differentiation of equations (9) and (10), respectively.

Let R be a region with boundary C and continuous functions m and w satisfy

$$\iint_R \left(\frac{\partial w}{\partial x_1} - \frac{\partial m}{\partial x_2}\right) dx_1 dx_2 = \int_C (m dx_1 + w dx_2)
 \tag{21}$$

The area $\bar{A} = \frac{1}{2} \int_{P_1}^{P_2} r^2 d\theta = \iint_R dx_1 dx_2$ of the domain R can be written over the boundary using the Green's theorem as [16-18]

$$\bar{A} = \frac{1}{2} \int_C (x_1 dx_2 - x_2 dx_1)
 \tag{22}$$

By discretizing the boundary of the structure into Q quadratic boundary elements, we have the following relation at c th element

$$x_m(\xi) = N^c(\xi)x_m^c
 \tag{23}$$

Also, the area can be expressed as follows

$$\bar{A} = \frac{1}{2} \sum_{b=1}^Q \int_{-1}^1 [x_1(\xi)n_1 + x_2(\xi)n_2]J(\xi)d\xi
 \tag{24}$$

where n_1 and n_2 can be written in terms of the Jacobian matrix of the transformation $J(\xi)$ as

$$n_1 = \frac{dx_2}{d\bar{A}} = \frac{dx_2/d\xi}{d\bar{A}/d\xi} = \frac{dx_2/d\xi}{J(\xi)} \quad (25)$$

$$n_2 = -\frac{dx_1}{d\bar{A}} = -\frac{dx_1/d\xi}{d\bar{A}/d\xi} = -\frac{dx_1/d\xi}{J(\xi)} \quad (26)$$

Substitution of equations (25) and (26) into equation (24) yields

$$\bar{A} = \frac{1}{2} \sum_{b=1}^q \int_{-1}^1 \left[x_1(\xi) \frac{dx_2}{d\xi} - x_2(\xi) \frac{dx_1}{d\xi} \right] d\xi \quad (27)$$

By differentiating (27) taking into consideration that

$$\frac{\partial}{\partial x_h} \left(\frac{dx_2(\xi)}{d\xi} \right) = 0 \quad (28)$$

and

$$\frac{\partial}{\partial x_h} (x_2(\xi)) = 0 \quad (29)$$

Therefore

$$\frac{\partial \bar{A}}{\partial x_h} = \frac{1}{2} \sum_{b=1}^q \int_{-1}^1 \left[\frac{\partial x_1(\xi)}{\partial x_h} \frac{dx_2}{d\xi} - x_2(\xi) \frac{\partial}{\partial x_h} \left(\frac{dx_1}{d\xi} \right) \right] d\xi \quad (30)$$

If x_h is the x_2 coordinate of a movable node, then

$$\frac{\partial}{\partial x_h} \left(\frac{dx_1(\xi)}{d\xi} \right) = 0 \quad (31)$$

and

$$\frac{\partial}{\partial x_h} (x_1(\xi)) = 0 \quad (32)$$

Therefore

$$\frac{\partial \bar{A}}{\partial x_h} = \frac{1}{2} \sum_{b=1}^q \int_{-1}^1 \left[x_1(\xi) \frac{\partial}{\partial x_h} \left(\frac{dx_2}{d\xi} \right) - \frac{\partial x_2(\xi)}{\partial x_h} \left(\frac{dx_1}{d\xi} \right) \right] d\xi \quad (33)$$

where weight minimization is equivalent to area minimization.

Now, we consider the following minimization problem

$$\text{Minimize} \quad \bar{A}(x_h) \quad (34)$$

$$\text{Subject to} \quad \chi_m(x_h) \leq 0, \quad m = 1, \dots, M \quad (35)$$

$$x_h^{\omega} \leq x_h \leq x_h^{\bar{\omega}} \quad (36)$$

where $x_h = [x_1, x_2, \dots, x_L]^T$.

The feasible direction method (FDM) can be successfully applied for solving the current optimization problem using the following iteration process:

$$x_h = x_{h-1} + s_h d_h \quad (37)$$

Under the following condition

$$\bar{A}(x_h) - \bar{A}(x_{h-1}) \leq \varepsilon \quad (38)$$

where h , ε , s_h and d_h are respectively iteration number, predefined tolerance, line step parameter, search direction d_h which can be defined as

$$d_h = -H^h \nabla \bar{A}(x_h) \quad (39)$$

where the inverse Hessian matrix can be approximated in terms of the identity matrix I by

$$H^{h+1} = \left[I - \frac{P^h Q^h}{(P^h)^T Q^h} \right] H^h \left[I - \frac{Q^h (P^h)^T}{(P^h)^T Q^h} \right] + \frac{P^h (Q^h)^T}{(P^h)^T Q^h} \quad (40)$$

In which

$$P^h = x_{h+1} - x_h, \quad Q^h = \nabla \bar{A}(x_{h+1}) - \nabla \bar{A}(x_h), \quad H^0 = I$$

Using FDM, we have

$$\nabla \bar{A}(x_h) d \leq 0 \quad (41)$$

and

$$\nabla \chi_m(x_h) d \leq 0 \quad (42)$$

Now, we want to solve the following search direction problem [19]

Maximize \mathcal{B}

$$\text{Subject to } d^T \nabla \chi_m(x_h) + \theta_m \mathcal{B} \leq 0 \quad (43)$$

$$d^T \nabla \bar{A}(x_h) + \mathcal{B} \leq 0 \quad (44)$$

$$-1 \leq d \leq 1 \quad (45)$$

where θ_m is the push-off factor which can be written as

$$\theta_m = \left[1 - \frac{\chi_m(x_h)}{\varepsilon} \right]^2 \theta_0 \quad (46)$$

where ε and θ_0 are constants.

We will use the preceding formulation when the design is inside the feasible domain. But when the design is outside the feasible domain, we will solve the following search direction problem

$$\text{Maximize } \nabla \bar{A}(x_h) \cdot d + \Phi \mathcal{B}$$

$$\text{Subject to } \nabla \chi_m(x_h) \cdot d + \theta_m \mathcal{B} \leq 0, \quad m \in J \quad (47)$$

$$d^T \cdot d \leq 1 \quad (48)$$

where J and Φ are respectively potential constraint set and weighting factor.

5. NUMERICAL RESULTS AND DISCUSSION

In order to illustrate the numerical results of the current study, the following physical constants for material "A" are as follows

Elasticity tensor

$$C_{ijkl} = \begin{bmatrix} 430.1 & 130.4 & 18.2 & 0 & 0 & 201.3 \\ 130.4 & 116.7 & 21.0 & 0 & 0 & 70.1 \\ 18.2 & 21.0 & 73.6 & 0 & 0 & 2.4 \\ 0 & 0 & 0 & 19.8 & -8.0 & 0 \\ 0 & 0 & 0 & -8.0 & 29.1 & 0 \\ 201.3 & 70.1 & 2.4 & 0 & 0 & 147.3 \end{bmatrix} \text{ GPa}$$

Mechanical temperature coefficient

$$\beta_{pj} = \begin{bmatrix} 1.01 & 2.00 & 0 \\ 2.00 & 1.48 & 0 \\ 0 & 0 & 7.52 \end{bmatrix} \cdot 10^6 \text{ N/Km}^2$$

Tensor of thermal conductivity

$$k_{pj} = \begin{bmatrix} 5.2 & 0 & 0 \\ 0 & 7.6 & 0 \\ 0 & 0 & 38.3 \end{bmatrix} \text{ W/Km}$$

Mass density $\rho = 7820 \text{ kg/m}^3$ and heat capacity $c = 461 \text{ J/kg K}$.

A prismatic material is taken as material **B** in the numerical calculations with the following physical constants

Elasticity tensor

$$C_{ijkl} = \begin{bmatrix} 60.23 & 18.67 & 18.96 & -7.69 & 15.60 & -25.28 \\ 18.67 & 21.26 & 9.36 & -3.74 & 4.21 & -8.47 \\ 18.96 & 9.36 & 47.04 & -8.82 & 15.28 & -8.31 \\ -7.69 & -3.74 & -8.82 & 10.18 & -9.54 & 5.69 \\ 15.60 & 4.21 & 15.28 & -9.54 & 21.19 & -8.54 \\ -25.28 & -8.47 & -8.31 & 5.69 & -8.54 & 20.75 \end{bmatrix} \text{ GPa}$$

Mechanical temperature coefficient

$$\beta_{pj} = \begin{bmatrix} 0.002 & 0.02 & 0.03 \\ 0.02 & 0.004 & 0.04 \\ 0.03 & 0.04 & 0.05 \end{bmatrix} \cdot 10^6 \text{ N/Km}^2$$

Tensor of thermal conductivity

$$k_{pj} = \begin{bmatrix} 0.8 & 0.1 & 0.15 \\ 0.1 & 0.9 & 0.12 \\ 0.15 & 0.12 & 0.7 \end{bmatrix} \text{ W/Km}$$

Mass density $\rho = 1600 \text{ kg/m}^3$ and heat capacity $c = 0.1 \text{ J/kg K}$.

Also, a monoclinic North Sea sandstone reservoir rock is taken as material **C** in the numerical computations with the following physical constants

Elasticity tensor

$$C_{ijkl} = \begin{bmatrix} 17.77 & 3.78 & 3.76 & 0.24 & 0.28 & 0.03 \\ 3.78 & 19.45 & 4.13 & 0 & 0 & 1.13 \\ 3.76 & 4.13 & 21.79 & 0 & 0 & 0.38 \\ 0 & 0 & 0 & 8.30 & 0.66 & 0 \\ 0 & 0 & 0 & 0.66 & 7.62 & 0 \\ 0.03 & 1.13 & 0.38 & 0 & 0 & 7.77 \end{bmatrix} \text{ GPa}$$

Mechanical temperature coefficient

$$\beta_{pj} = \begin{bmatrix} 0.001 & 0.02 & 0 \\ 0.02 & 0.006 & 0 \\ 0 & 0 & 0.05 \end{bmatrix} \cdot 10^6 \text{ N/Km}^2$$

Tensor of thermal conductivity

$$k_{pj} = \begin{bmatrix} 1 & 0.1 & 0.2 \\ 0.1 & 1.1 & 0.15 \\ 0.2 & 0.15 & 0.9 \end{bmatrix} \text{ W/Km}$$

Mass density $\rho = 2216 \text{ kg/m}^3$ and heat capacity $c = 0.1 \text{ J/kg K}$.

Table. 1. Optimization analysis for considered materials

Material	Iterations	Percentage change between final and initial value	Maximum stress	Reduction of compliance
A	12	65%	0.411	92.40
B	12	65%	0.390	90.87
C	12	73%	0.223	91.10

Table 2. Comparison of computer resources needed for FEM and DRBEM modelling of the right half of the link plate design

	FEM	DRBEM
Number of elements	12980	48
CPU-Time [min.]	190	3
Memory [Mbyte]	140	0.6
Disc space [Mbyte]	200	0
Accuracy of results [%]	2.3	1.3

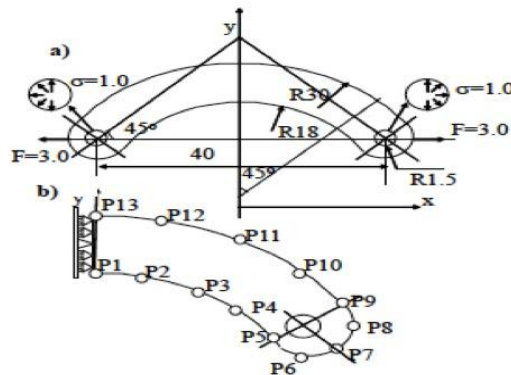


Fig. 1. a) Geometry, b) Boundary element model for the link plate.

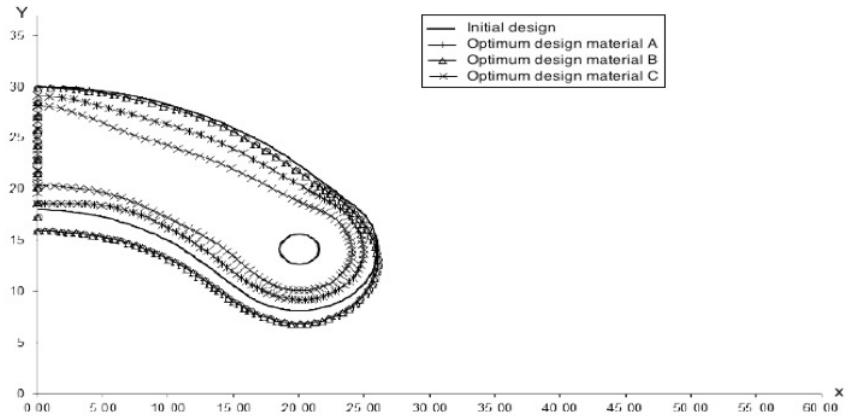


Fig. 2. Optimum shape design for the link plate.

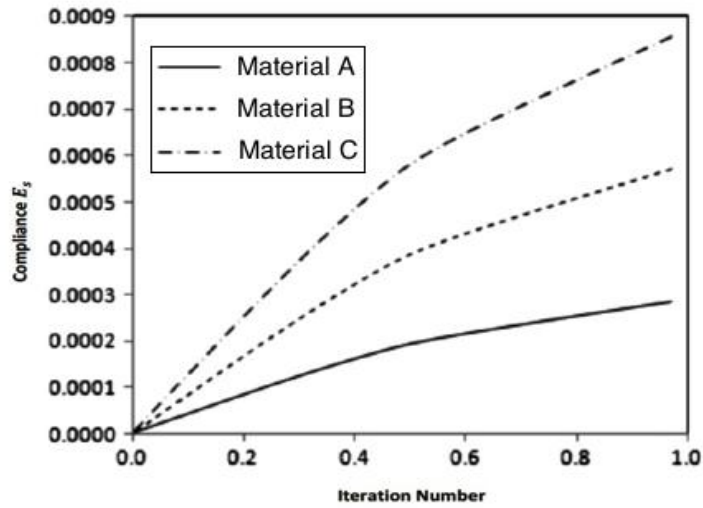


Fig. 3. Compliance iteration history for the link plate.

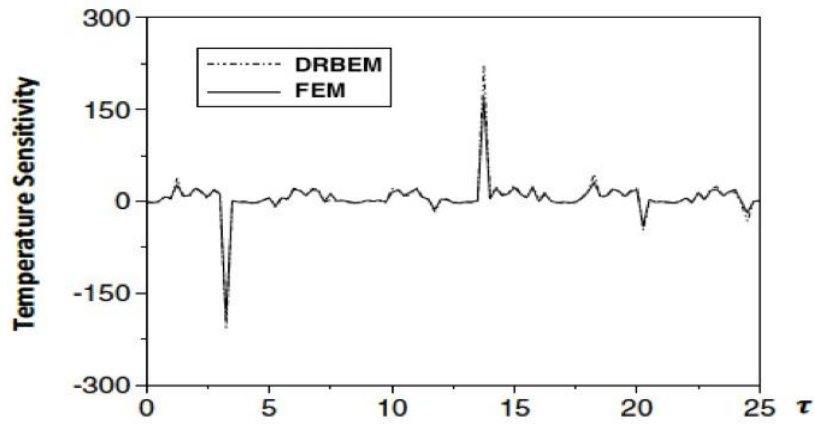


Fig. 4. Variation of the temperature T sensitivity with time τ .

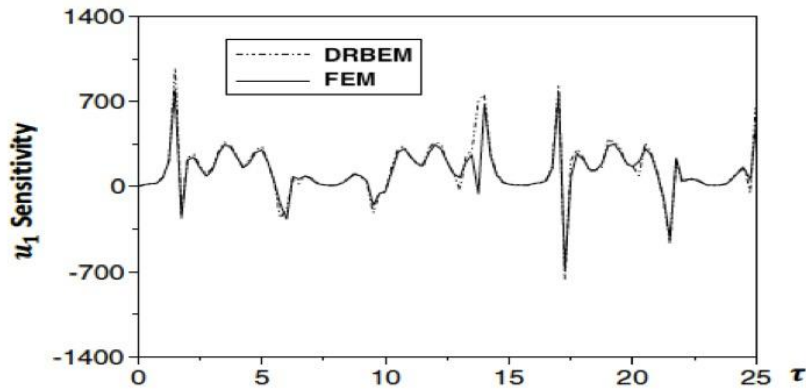


Fig. 5. Variation of the displacement u_1 sensitivity with time τ .

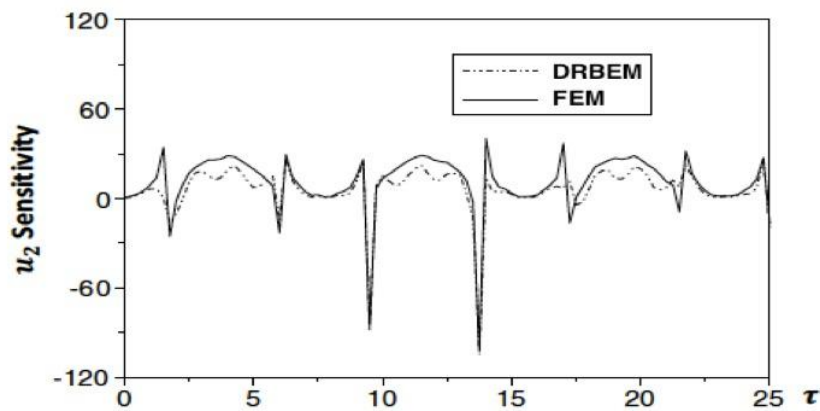


Fig. 6. Variation of the displacement u_2 sensitivity with time τ .

For the purpose of numerical calculations of materials A, B and C, we considered the following constants

$$H_0 = 1000000 \text{ Oersted}, \mu = 0.5 \text{ Gauss/Oersted}, \tau_0 = 0.5, m = 0.5, \Delta\tau = 0.0001, Q_0 = 0.5, \nu = 1.$$

It can be noticed from numerical results that the DRBEM results are in very good agreement with those obtained using the finite element method (FEM) of Gao and Yao [23]. The right half of the link plate shown in Fig. 1a and design variables shown in Fig. 1b are considered. The optimum shapes of the considered structure for selected anisotropic materials produced from the current study are shown in Fig. 2. It can be seen that the weight and the maximum stress have increased. Fig. 3 shows the iteration history for elastic compliance of the link plate for selected materials A, B and C. It can be seen that the weight and the maximum stress of the link plate have been decreased (see Table 1). Figs. 4 and 5 show the

sensitivities of the displacement distributions. Also, Fig. 6 shows the sensitivity of the temperature distribution to demonstrate the accuracy of the current technique (see Table 2). For further finite difference method details, we refer the reader to many researchers [24-29]. Also, for more boundary element method details we refer the reader to many researchers [30-55].

6. CONCLUSION

Our results thus confirm the validity, accuracy, and efficiency of the proposed technique. It is noted that the obtained dual reciprocity boundary element method (DRBEM) results are more accurate than the FEM results, the DRBEM is more efficient and easy to use than FEM because it only needs the boundary of the domain needs to be discretized.

COMPETING INTERESTS

Authors have declared that no competing interests exist.

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